

LAGRANGE

ESA's Space Weather Mission to L5

Stefan Kraft on behalf of the LAGRANGE team
Mission Study Manager
Space Weather Office
Space Safety Programme Office

21/11/2019

Enhanced Space Weather Monitoring System



LAGRANGE Missions

L5



Forecasting
&
Event detection

L1

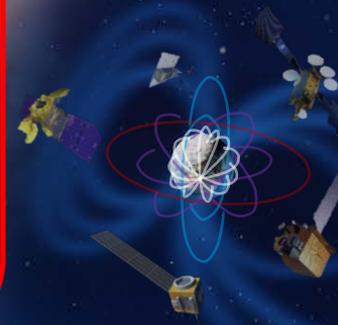


D3S

(Distributed SWE
Sensor System)

Impact & State
monitoring

Hosted payloads



Objectives of ESA's SWE Monitoring System



Establish observational system that continuously monitors space weather

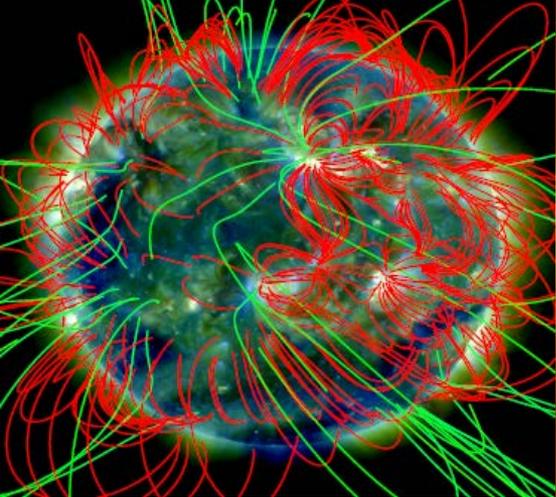
- From the Lagrangian points L1 and L5 (collaboration with US – NOAA/NASA, others ...)
- In the proximity of the Earth (LEO, MEO, GEO)
- Ground based measurements

Advantages of L5 position

- Monitoring of space between Earth and Sun from a **side view**
- Increases our visibility of the Sun to **2/3 of the Sun's surface**
- Activities on the Sun are seen about **4.5 days earlier** than from Earth

Some specifics of an observatory at L5

- **Short latencies** (15 min to 1 h) and continuous operation (**24/7**) to keep track of changes
- High reliability (0.75 @ 7.5 years) and **high system availability** (99%)
- Requires the development approach of an operational mission
- **Science enabling** through observations and 'high data rate' when possible
- Good heritage from science missions (SOHO, STEREO, Solar Orbiter, Cluster, etc)
- First mission has the character of an Explorer (**first of its kind**)



EUV Imager (EUVI)

Single channel 19.5 nm

Photospheric Magnetic Field Imager (PMI)

Vector magnetograph (Zeeman line scanning)

Compact Coronagraph (CCOR)

~3 to 22 solar radii

Heliospheric Imager (HI)

4 to 70 degree

LAGRANGE (L5)

Remote/optical

X-ray Flux Monitor (XFM)

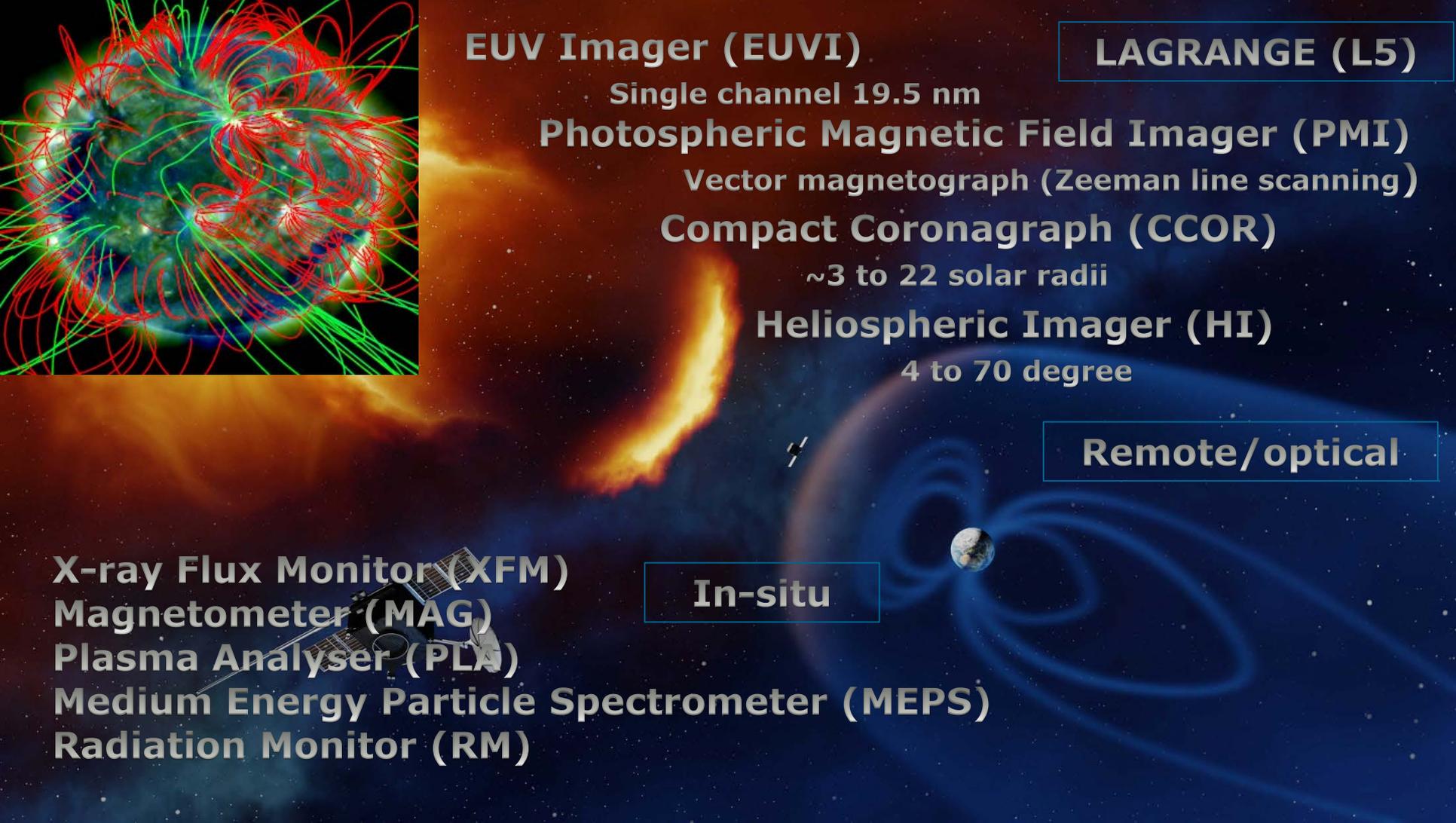
Magnetometer (MAG)

Plasma Analyser (PLA)

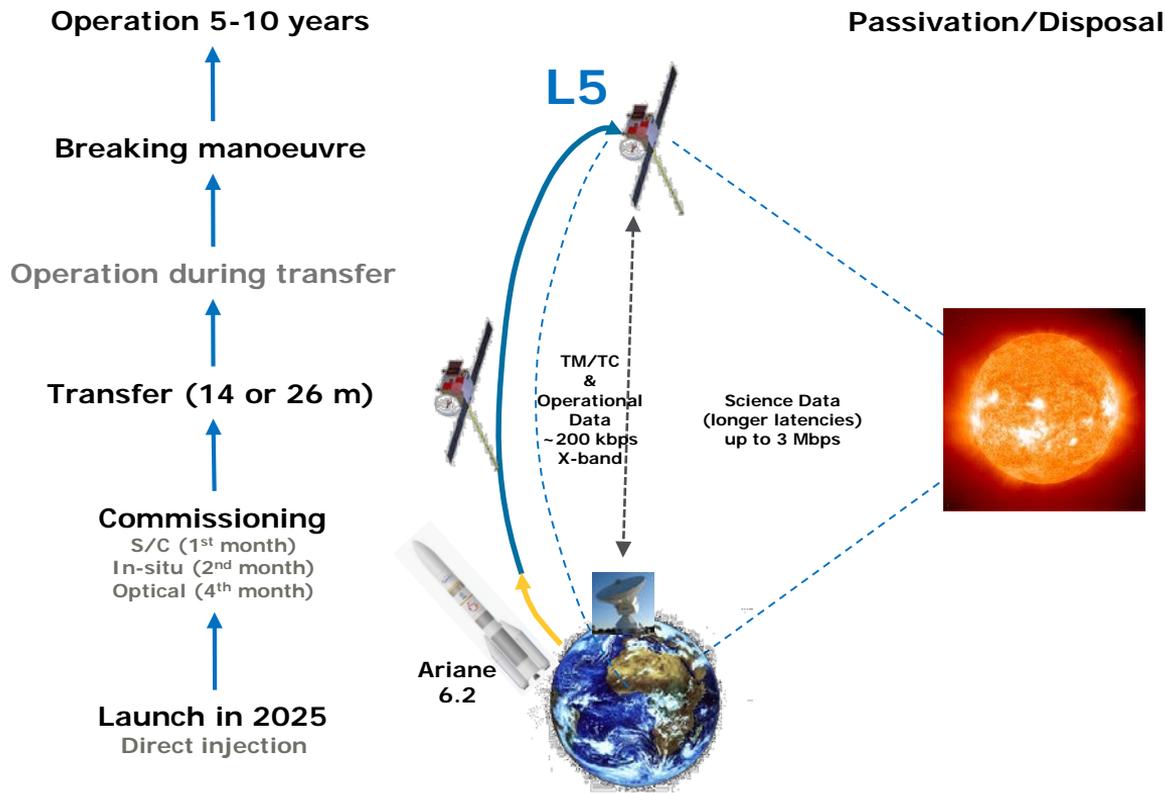
Medium Energy Particle Spectrometer (MEPS)

Radiation Monitor (RM)

In-situ



Mission architecture



24/7 operation using ESTRACK



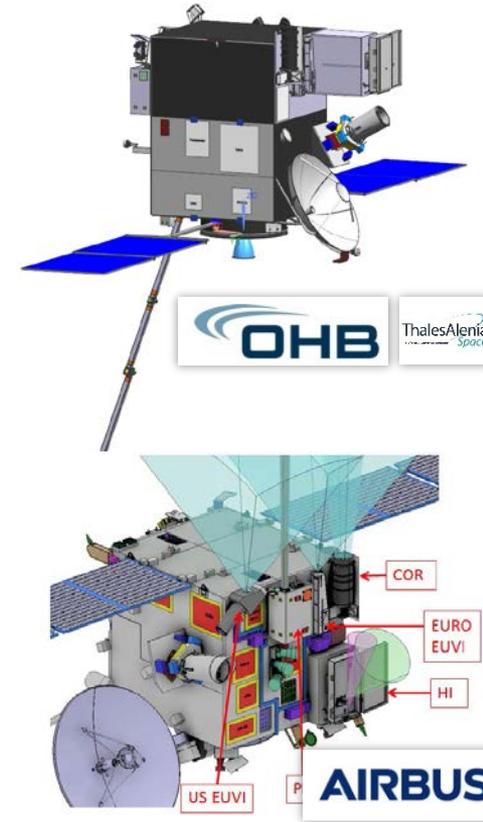
Data transmission via flexible file based data handling



L5 space weather observatory key features



| Feature | Value | Comment |
|--|--|---|
| Mass, power | ~2 t wet mass, <1.5 kW | Propellant large fraction |
| Delta-v | Between ~1100 and 1800 m/s | One or two revolutions |
| Dimensions | 2.5 m x 2.5 m x 2.5 m | Medium sized S/C |
| Launcher | Ariane 6.2 | Single passenger |
| Instruments | 4 optical (200 kbps) 5 in-situ (10 kbps) | 120 kg, 200 W |
| Magnetic cleanliness | Optimised for magnetometer | ~5nT at sensor on boom |
| 'Silent' observations 3-axes-stabilised | Pointing stability | Imaging performance |
| Clean environment | Thruster configuration optimised for observations | Plasma measurements |
| Continuous observation | 24/7 - short latencies (as short as 15 minutes) | Also during Carrington event – optimisation |
| Orbit | Lagrange Point 5 | 1 AU distance from Earth ~1.5 Mkm eccentricity |



LAGRANGE instruments



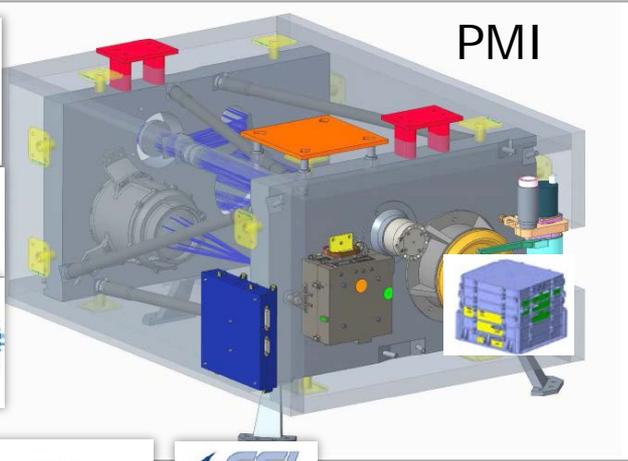
| Instrument | Observation | Utilisation | Performance (latency/cadence) |
|--|--|--|--|
| Photospheric Magnetic field Imager (PMI) | Vector magnetic field mapping of the solar photosphere | Evolving magnetic complexity: input into solar wind modelling and activity forecast | Range: +/-4 kG Accuracy: 10 G, Zero 0.1G 1.2 arcsec / pixel, (40'/ 30') |
| EUV Imager (EUVI) | EUV solar disk imagery | Monitor solar active regions, coronal holes and on-disk signatures | 19.5 nm 1.6 arcsec / pixel, (15'/5') |
| Compact Coronagraph (CCOR) | Solar coronagraphy | Evolution and propagation of CMEs | 3 to 22 Solar Radii 1 arcmin / pixel, (15'/5') |
| Heliospheric Imager (HI) | Heliospheric imagery | | 4° to 70°, 2 arcmin / pixel, (50'/30') |
| Plasma Analyser (PLA) | Solar wind particle densities, temperatures and velocity | Solar wind monitoring, detection and characterisation of high-speed solar wind streams | ±22.5° x 45°, 5° angular 70 eV to 33 keV, (15'/1') |
| Medium Energy Particle Spectrometer (MEPS) | Medium energetic particles | | 30 kev/nuc up to 8 MeV/nuc (ions) 30 keV to 0.6 MeV(electrons) 32/16 channels, 2x2 look directions, (15",1') |
| Magnetometer (MAG) | Interplanetary Magnetic Field vector-magnetic field | | Absolute accuracy 1 nT (15'/1") |
| Radiation Monitor (RM) | Highly energetic particles | Solar energetic particle event detection and monitoring | 2 MeV to 1 GeV (p+, ions), directional 100 keV to 8 MeV (electrons), (15'/1') |
| X-ray Flux Monitor (XFM) | Solar disk integrated X-ray flux | Detection and categorisation of solar flares | 0.05 to 0.8 nm / 1.55 keV to 25 keV 512 channels, 400 eV @ 6 keV, (15'/1') |



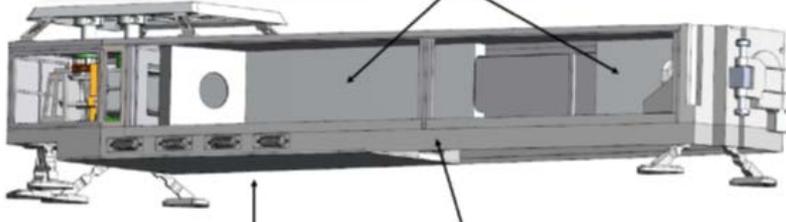
Remote Sensing Instruments



Poster J. Davies



PMI

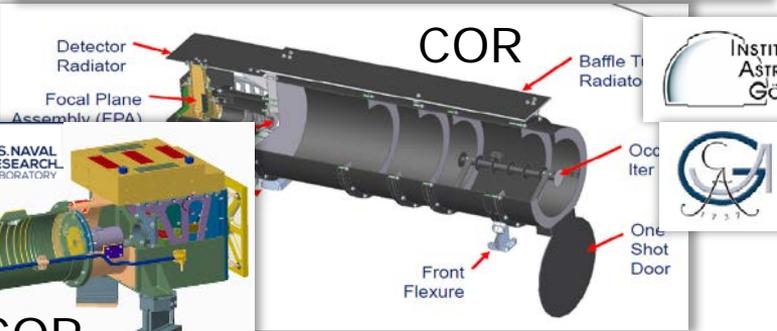


EUVI Talk by M. West

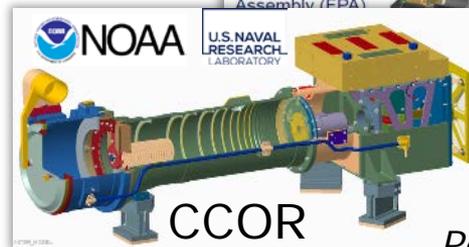


HI

CCOR



COR



CCOR

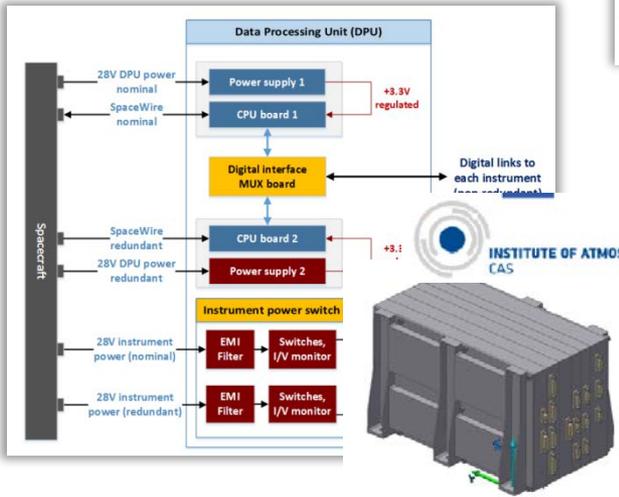
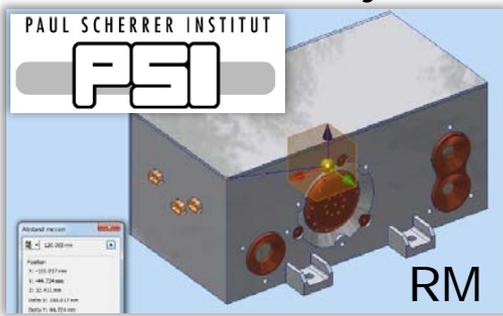
Posters J. Davies / J. Tappin



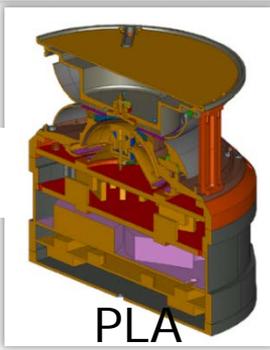
In-situ Instruments



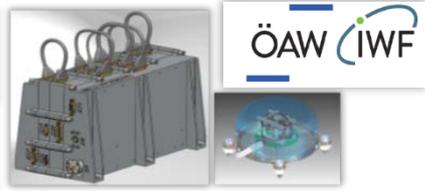
Talk W. Hajdas



UNIVERSITY COLLEGE LONDON
DEPARTMENT OF SPACE & CLIMATE PHYSICS
MULLARD SPACE SCIENCE LABORATORY



Talk
D. Kataria
Poster
J. Rae



Instrument to be procured individually

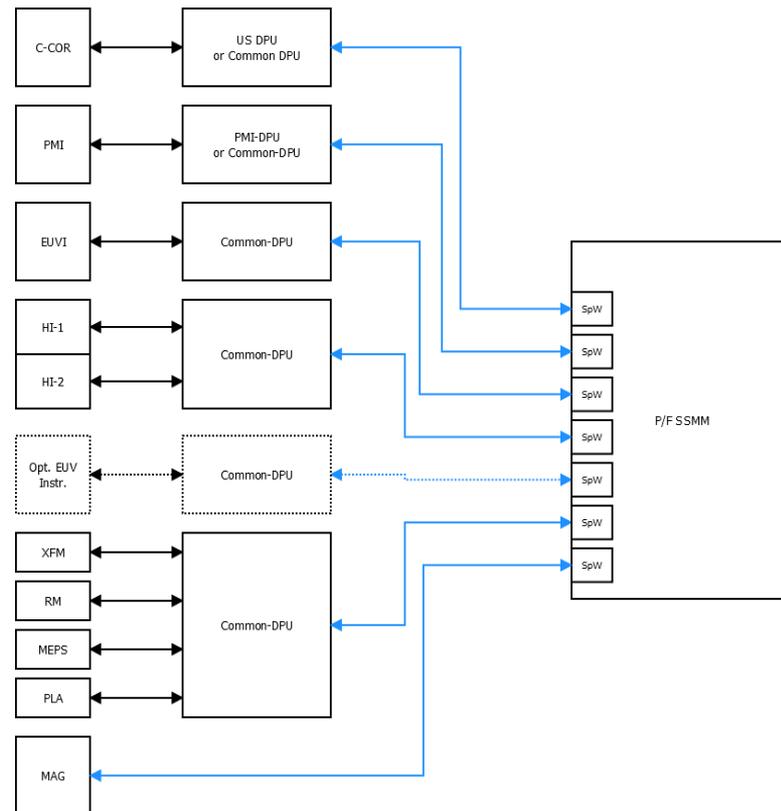


Data Handling System

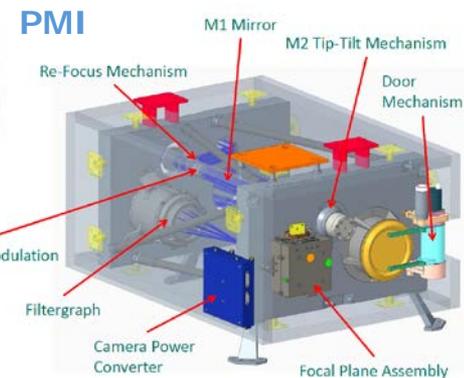
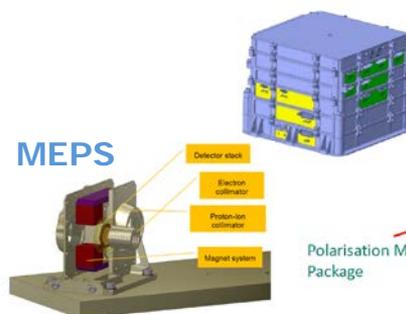
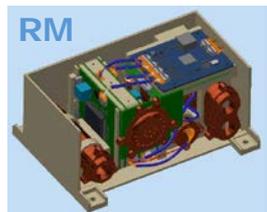
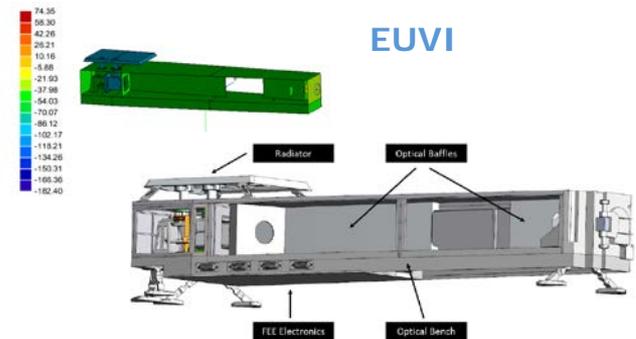
Internal re-assessment performed at ESA
 3 options recommended for further evaluation
 by industry

| Option | Description | Trade-off result | Comment |
|--------|---|------------------|---|
| #1-A | Baseline configuration at ISRR (IPCU - partially centralised) | Recommended | Trade-off result from Phase O/A study. |
| #3-A | Common ICU+DPU design for all remote sensing instruments, with no central processing function. | Recommended | Typical architecture used in Science missions (JUICE evolution). |
| #3-B | Common I-CDPU design for all remote sensing instruments and in-situ instrument suite, with a centralized unit for image processing and compression (e.g. OBC). | Recommended | As #3-A Processing partly performed by platform |

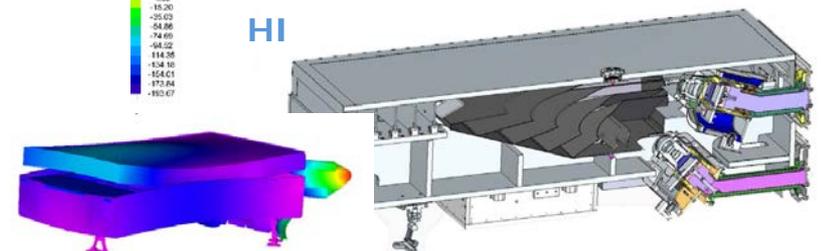
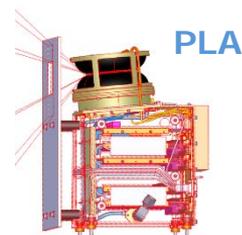
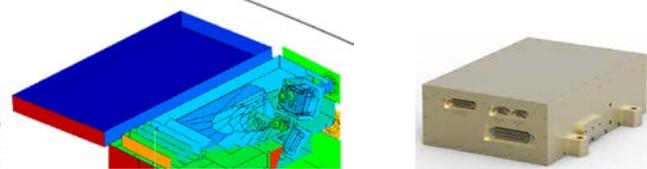
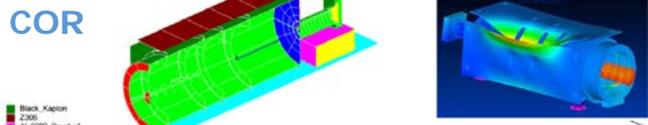
Distributed system anticipated



Lagrange mission instrument status



Good progress in all design areas



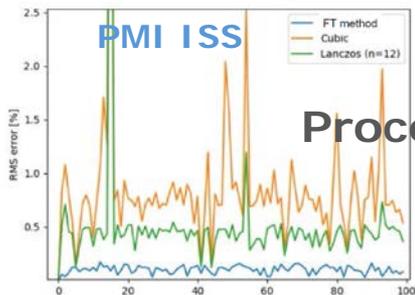
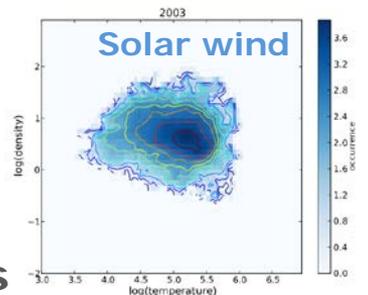
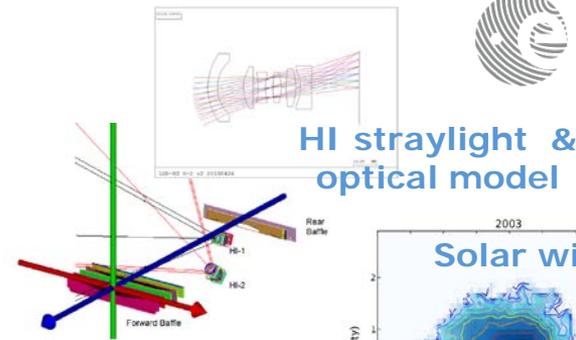
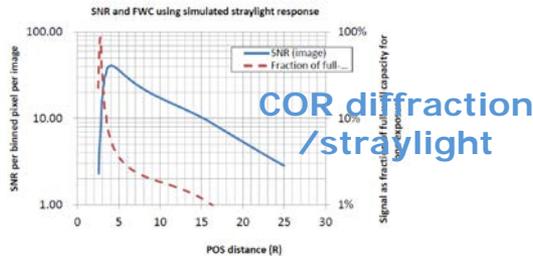
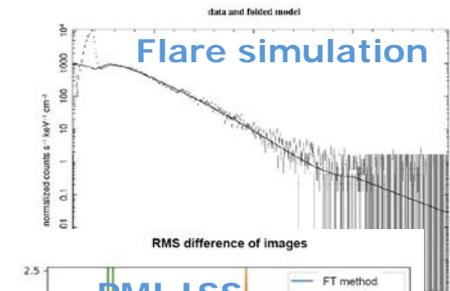
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Stefan Kraft | ESOC | LGR Status ESWW 16 | 21/11/2019 | Slide 11



European Space Agency

Instruments performance status

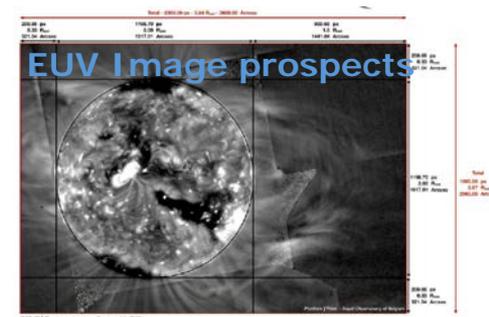
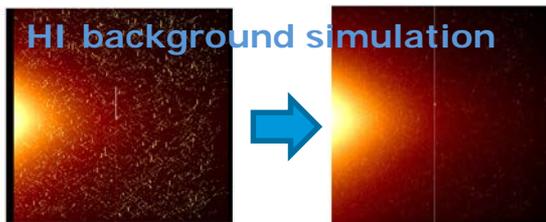


Performance modelling

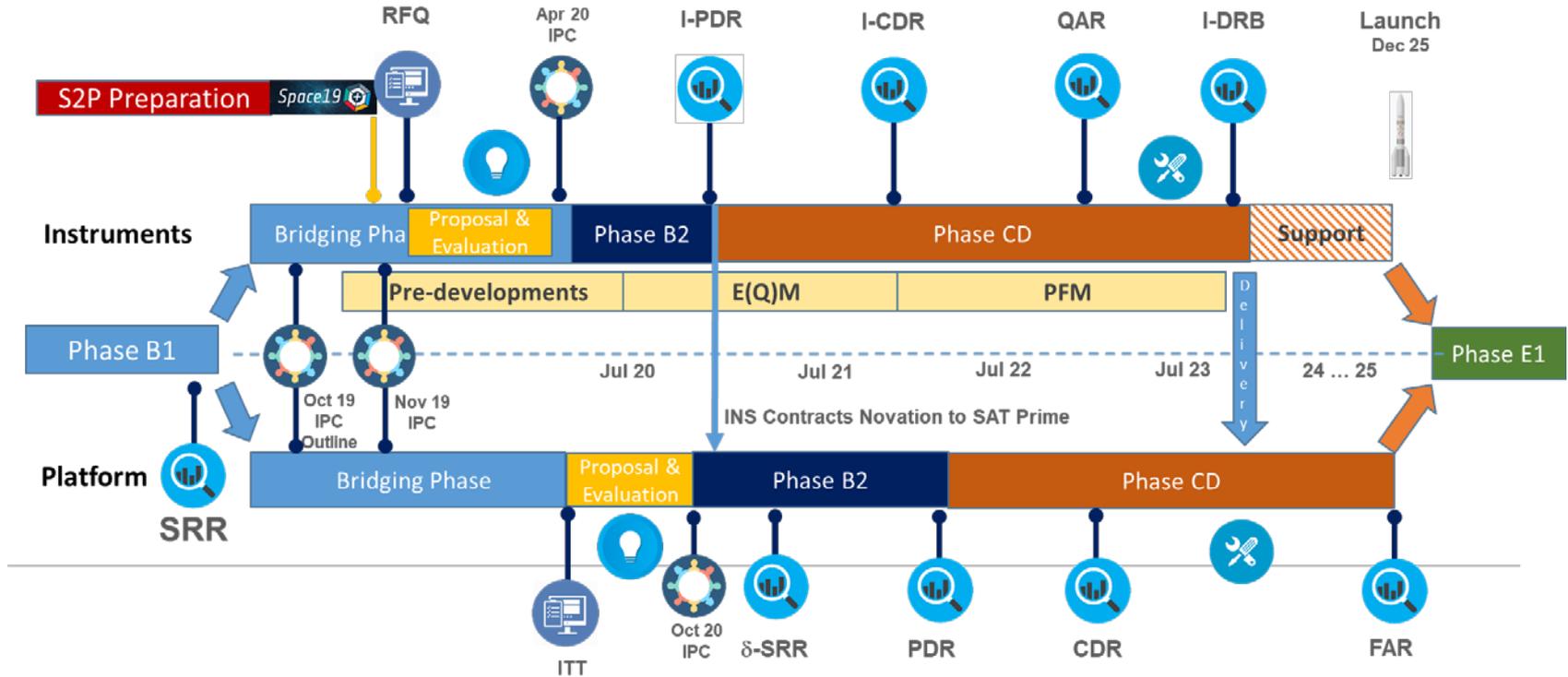
Processing simulations & Requirements analysis

Poster I. Grozea

First hardware coming



LAGRANGE Mission development schedule





THANK YOU

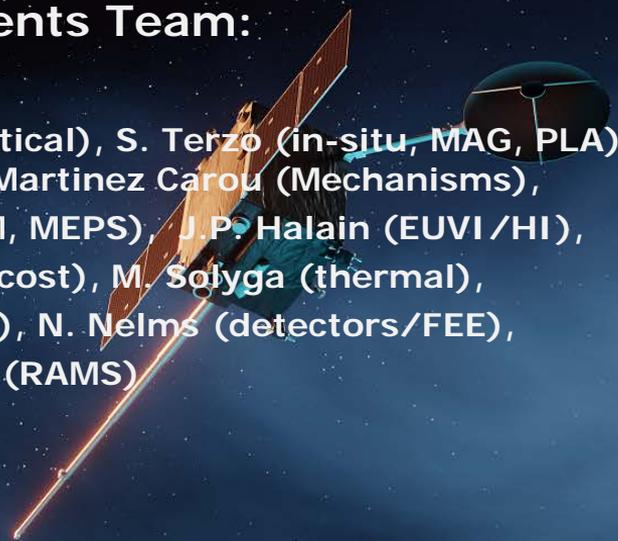
&

ESA LAGRANGE Instruments Team:

**F. Ratti (instrument coordination), I. Biswas (PMI, optical), S. Terzo (in-situ, MAG, PLA),
A. Lupi (I/F), A. Baselga Mateo (EUVI/COR), A. Martinez Carou (Mechanisms),
D. Rodgers (PLA), D. Steenari (DHS), G. Santin (RM, MEPS), J.P. Halain (EUVI/HI),
J.P. Luntama (mission scientist), L. Visconti (cost), M. Solyga (thermal),
N. Bozic (structures), P. Jiggins (environment), N. Nelms (detectors/FEE),
R. Franco (E2ES), V. Favia (RAMS)**

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Other talks and posters



| | |
|--|----------------------------------|
| The EUV Imager on Lagrange | <i>West, M et al. (talk)</i> |
| In-situ particle instruments for enhanced space-weather monitoring | <i>Kataria, D et al. (talk)</i> |
| Design, development and testing of the RADEM and NGRM instruments | <i>Hajdas, W et al. (talk)</i> |
| The Remote-Sensing Package for the Lagrange Mission | <i>Davies, J et al. (poster)</i> |
| The COR and HI Instruments for the Lagrange Mission | <i>Davies, J et al. (poster)</i> |
| Space weather monitoring of the in-situ environment from the Sun-Earth Lagrange points | <i>Rae, J et al. (poster)</i> |
| LGR-RS End to End Performance Simulator Architectural Design and First Results | <i>Grozea, I et al. (poster)</i> |

